Analysis and design of oscillatory control systems

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Abstract

This paper presents analysis and design results for control systems subject to oscillatory inputs, i.e., inputs of large amplitude and high frequency. The key analysis results are a series expansion characterizing the averaged system and various Lie-algebraic conditions that guarantee the series can be summed. Various example systems provide insight into the results. With regards to design, we recover and extend a variety of point stabilization and trajectory tracking results using oscillatory controls. We present novel developments on stabilization of systems with positive trace and on tracking for second order underactuated systems.

Keywords

oscillatory control, averaging, geometric methods, point stabilization, trajectory tracking for underactuated systems

I. INTRODUCTION

The paper investigates the behavior of finite dimensional analytic systems subject to oscillatory controls. We present averaging analysis and control design results for systems described by a differential equation of the form

$$\dot{x}=f(t,x)+rac{1}{\epsilon}g\left(rac{t}{\epsilon},t,x
ight)\,,$$

where the vector field g is periodic in its first argument, ϵ is a small positive parameter, and both vector fields f and q are analytic in x. Our objective is to provide a rigorous

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Jorge Cortés and Francesco Bullo are with the Coordinated Science Laboratory and the General Engineering Department, University of Illinois at Urbana-Champaign, 1308 W. Main St., Urbana, IL 61801, United States, Tels: +1-217-244-8734 and +1-217-333-0656, Fax: +1-217-244-1653, Email: {jcortes,bullo}@uiuc.edu and general framework that allows to obtain (1) a coordinate-free expression of the averaged system, and a series expansion representation for it; (2) control design tools for point stabilization and trajectory planning in underactuated systems.

Motivation

The study of oscillations in nonlinear differential equations is a classic and widespread research topic. Related research areas include nonlinear dynamical systems [1], nonlinear and geometric control [2], [3], analysis of animal locomotion [4], design of robotic locomotion and manipulation devices [5], analysis of switching circuit models and power conversion circuits [6], control of quantum dynamics [7] and chemical reactions [8], analysis, design, and control of biomineralization and crystallization processes [9], [10], and so forth.

Furthermore, averaging analysis seems well suited to tackle novel applications in the field of micro-electro mechanical systems and vibrational control is being investigated within the context of active control of fluids and separation control. Examples include [11] on the scale dependence in oscillatory control of mechanical systems, [12] on unsteady flow and separation control using oscillatory blowing.

From a control theoretical viewpoint, we study oscillatory controls for the purpose of stabilization and tracking problems. For classes of nonlinear underactuated systems, it is interesting to investigate what control objectives can be obtained via the use of high frequency, high amplitude inputs. Since modern textbooks [13], [14], [15], [16] do not present a comprehensive approach on perturbation methods in control theory, we endeavor to develop novel tools and shed further light onto these problems.

Literature review

This work has connections with classic averaging theory (see [17] for a standard treatment), as well as with numerous ongoing research efforts. First of all, our analysis complements the study of differential equations subject to periodic high frequency, high amplitude forcing terms; see [18], [19], [20]. In these works, the coupling effect between the input vector fields plays a key role: typically, Lie brackets between them appear, and in the averaging approximation the trajectories of the original system converge to those of the averaged system. Here we shall focus our attention on systems where the interaction takes place between the drift and the input vector fields.

A second set of related results deals with the analysis of high frequency vibrations in mechanical and other types of systems [21], [22], [23], [24], [25], [26], [27], [28], and more generally averaging analysis in locomotion, rectification and other physical phenomenon where non-commuting vector fields play a role; see [29], [30], [31].

Finally, there are three related areas within the context of control design. One of these is the design of time-varying stabilizing laws for driftless systems, (sometimes referred to as nonholonomic), see for instance [32], [33], [34], [35], [36]. A second area deals with the design of oscillatory controls for point stabilization in general nonlinear and mechanical control systems; see [37], [38], [39], [2], and a third area is devoted to the design of oscillatory controls for trajectory planning in driftless systems [40], [41], and for constructive controllability and approximate inversion [42], [43].

Statement of contributions

The first contribution of this paper is a general averaging analysis in a coordinate-free differential geometric setting. We give a novel sufficient condition for general nonlinear systems based on the commutativity of the input vector fields which enables us to perform the averaging procedure. Exploiting a generalized variation of constants formula, we provide a new explicit representation of the averaged system for analytic control systems with two time scales. This representation consists of an infinite sum of Lie brackets of the input vector fields with the drift and iterated integrals of the open-loop controls. Finally, we particularize our discussion to various classes of systems including bilinear, Hamiltonian, and second order systems, extending a number of previous results on approximate descriptions and obtaining new sufficient conditions that guarantee the series for the averaged system is summable.

After completing this general analysis, we present various design tools and results for vibrational control. Regarding point stabilization, we provide sufficient conditions for the existence of an equilibrium point for the averaged system, we prove that the order of linearizing and averaging is non-influential, and we design oscillatory controls to stabilize the averaged systems. In particular, we recover the known result on stabilization of systems with negative linearization trace, and prove a novel result on stabilization of systems with positive trace (via nonlinear feedback). Regarding trajectory tracking, we exploit our analysis results on nonlinear systems with two time scales to steer the averaged system along arbitrary reference paths. We focus on second order underactuated systems and develop a novel controller using oscillatory signals to track a desired smooth trajectory. We apply the strategy to a second-order nonholonomic integrator and to the PVTOL system.

Organization

We introduce some preliminary concepts in Section II. Section III presents the main averaging analysis, and Section IV treats various classes of systems for which the series expansions assume a particular structure. Section V and Section VI discuss respectively stabilization and tracking via oscillatory controls. Finally, we present our conclusions in Section VII.

II. PRELIMINARIES AND NOTATION

This section contains some basic definitions and results on iterated integrals of scalar functions and on differential geometry.

A. Iterated integrals and their averages

Let \mathbb{N} be the set of strictly positive integers and $\mathbb{R}_+ = [0, +\infty)$. Let \mathcal{I} be the set of all nontrivial multiindices $I = (i_1, \ldots, i_k)$, where i_1, \ldots, i_k take values in $\{1, \ldots, m\}$. Given m bounded measurable functions $u_i : \mathbb{R}_+ \to \mathbb{R}$, define their *iterated integrals* $\{U_I : \mathbb{R}_+ \to \mathbb{R}, I \in \mathcal{I}\}$ by

$$U_{(i_1,...,i_k)}(t) = \int_0^t u_{i_k}(t_k) \int_0^{t_k} u_{i_{k-1}}(t_{k-1}) \, \dots \int_0^{t_2} u_{i_1}(t_1) dt_1 \dots dt_k \, .$$

Let S be a set of $k_1 + \cdots + k_m$ elements. Let $C_{k_1,\ldots,k_m}(S)$ denote the collection of all possible ways of taking m classes of members of S, with the *i*th class having k_i elements. The cardinality of $C_{k_1,\ldots,k_m}(S)$ is the multinomial coefficient

$$egin{pmatrix} k_1+\dots+k_m\ k_1,\dots,k_m \end{pmatrix} = rac{(k_1+\dots+k_m)!}{k_1!\dots k_m!}$$

To each element $\alpha \in C_{k_1,...,k_m}(S)$, we associate a multiindex $I(\alpha)$ of length $k_1 + \cdots + k_m$ as follows: as $i \in \{1, ..., m\}$, place the index i in the k_i places corresponding to the ith class of α .

Given m bounded measurable functions $u_i: \mathbb{R}_+ \to \mathbb{R}$, define their multinomial iterated integrals $\{\mathcal{U}_{k_1,...,k_m}: \mathbb{R}_+ \to \mathbb{R}, k_1, \ldots, k_m \in \mathbb{N}\}$ according to

$$\mathcal{U}_{k_1,\ldots,k_m}(t) = \sum_{\alpha \in C_{k_1,\ldots,k_m}(S)} U_{I(\alpha)}(t) \,. \tag{1}$$

Furthermore, let $\mathcal{U}_{0,...,0}(t) \equiv 1$.

Lemma II.1: Let u_1, \ldots, u_m be bounded measurable functions. Their multinomial iterated integrals satisfy

$$\mathcal{U}_{k_1,\ldots,k_m}(t)=rac{1}{k_1!\ldots k_m!}\left(\int_0^t u_1(au)d au
ight)^{k_1}\ldots\left(\int_0^t u_m(au)d au
ight)^{k_m}\,.$$

The functions $\mathcal{U}_{k_1,\ldots,k_m}$ are *T*-periodic if and only if u_1,\ldots,u_m are *T*-periodic and zeromean.

Proof: We prove the result by induction on $k = k_1 + \cdots + k_m$. For k = 1, we have that $k_i = \delta_{ij}$, for some $j \in \{1, \ldots, m\}$. Then, $\mathcal{U}_{k_1, \ldots, k_m}(t) = \int_0^t u_j(\tau) d\tau = U_{(j)}(t)$.

Assume the claim is true for k - 1 and let us prove it for k. Using the induction hypothesis, the derivative of the right-hand side of eq. (2) can be written as,

$$egin{aligned} &rac{1}{k_1!\ldots k_m!} \left(k_1 u_1 U_{(1)}^{k_1-1}\ldots U_{(m)}^{k_m}+\cdots+k_m u_m U_{(1)}^{k_1}\ldots U_{(m)}^{k_m-1}
ight) \ &= u_1(t)\mathcal{U}_{k_1-1,\ldots,k_m}(t)+\cdots+u_m(t)\mathcal{U}_{k_1,\ldots,k_m-1}(t)\,, \end{aligned}$$

where $\mathcal{U}_{l_1,...,l_m}(t)\equiv 0$ if any of the l_i is negative. Integrating with respect to time, we obtain

$$\int_0^t \Big(u_1(s)\mathcal{U}_{k_1-1,...,k_m}(s)+\dots+u_m(s)\mathcal{U}_{k_1,...,k_m-1}(s)\Big) ds \;.$$

The claim now follows by noting that this equation is equivalent to the definition of multinomial iterated integral (1).

Next, we prove the second statement. All functions $\mathcal{U}_{k_1,\ldots,k_m}$ are *T*-periodic if and only if all functions $\int_0^t u_i(\tau) d\tau$ are *T*-periodic. Since

$$\int_0^{T+t} u_i(au) d au = \int_0^T u_i(au) d au + \int_T^{T+t} u_i(au) d au,$$

the functions $\int_0^t u_i(\tau) d\tau$ are *T*-periodic if u_i are zero-mean and *T*-periodic. Furthermore, the u_i are *T*-periodic if their time integrals are *T*-periodic, and they are zero-mean if $\int_0^T u_i(\tau) d\tau = \int_0^0 u_i(\tau) d\tau = 0.$

In the single-input case, m = 1, $u_1 = u$, we have $\mathcal{U}_k(t) = \sum_{\alpha \in C_k} U_{I(\alpha)}(t) = U_{(1,\dots,1)}(t)$, which we will simply denote by $U_k(t)$. Note that $U_2(t)$ and $U_{(2)}(t)$ denote different functions.

Given a T-periodic function V(t), let us denote

$$\overline{V} = rac{1}{T}\int_0^T V(t) dt$$
 .

Let $\|V\|_{\mathcal{L}_{\infty}}$ be the supremum of the absolute value V(t) for all $t \in \mathbb{R}_+$. Note that, since V is T-periodic,

$$\left\|V
ight\|_{\mathcal{L}_{\infty}} = \sup_{t\in [0,T]} \left|V(t)
ight|$$
 .

Lemma II.2: Let u_1, \ldots, u_m be bounded measurable, T-periodic and zero-mean functions. Then

$$egin{aligned} &\|\mathcal{U}_{k_{1},...,k_{m}}\|_{\mathcal{L}_{\infty}} \leq rac{T^{k_{1}+...+k_{m}}}{k_{1}!\ldots k_{m}!} \,\|u_{1}\|_{\mathcal{L}_{\infty}}^{k_{1}}\ldots \|u_{m}\|_{\mathcal{L}_{\infty}}^{k_{m}} \;, \ &|\overline{\mathcal{U}}_{k_{1},...,k_{m}}| \leq rac{T^{k_{1}+...+k_{m}}}{k_{1}!\ldots k_{m}!(1+\sum_{j=1}^{m}k_{j})} \,\|u_{1}\|_{\mathcal{L}_{\infty}}^{k_{1}}\ldots \|u_{m}\|_{\mathcal{L}_{\infty}}^{k_{m}} \;. \end{aligned}$$

Proof: Recall that, under the given assumptions, $\mathcal{U}_{k_1,...,k_m}(t)$ are T-periodic. For $0 \leq t \leq T,$

$$egin{aligned} |\mathcal{U}_{k_1,...,k_m}(t)| &\leq rac{1}{k_1!\ldots k_m!} \left(\int_0^t |u_1(au)|d au
ight)^{k_1}\ldots \left(\int_0^t |u_m(au)|d au
ight)^{k_m} \ &\leq rac{1}{k_1!\ldots k_m!} t^{k_1}\,\|u_1\|_{\mathcal{L}_\infty}^{k_1}\ldots t^{k_m}\,\|u_1\|_{\mathcal{L}_\infty}^{k_m} \,\,, \end{aligned}$$

which gives the first bound. The second one is proven via the chain of inequalities

$$|\overline{\mathcal{U}}_{k_1,...,k_m}| \leq rac{1}{T}\int_0^T |\mathcal{U}_{k_1,...,k_m}(t)| dt \leq rac{\|u_1\|_{\mathcal{L}_\infty}^{k_1}\dots\|u_1\|_{\mathcal{L}_\infty}^{k_m}}{k_1!\dots k_m!}rac{1}{T}\int_0^T t^{k_1+...+k_m} dt\,.$$

As an example, consider the functions $u_i(t) = a_i \cos \omega t$, $\omega \in \mathbb{N}$. Then,

$$\mathcal{U}_{k_1,...,k_m}(t) = rac{a_1^{k_1}\ldots a_m^{k_m}}{k_1!\ldots k_m!} \left(rac{1}{\omega}\sin\omega t
ight)^{k_1+\dots+k_m}$$

From the identity $4^m \int_0^{2\pi} (\sin t)^{2m} dt = 2\pi \begin{pmatrix} 2m \\ m \end{pmatrix}$ in [44], the averages are

$$\overline{\mathcal{U}}_{k_1,\dots,k_m} = \begin{cases} 0 & \text{if } k \text{ is odd} \\ \frac{a_1^{k_1}\dots a_m^{k_m}}{k_1!\dots k_m!} \left(\frac{1}{2\omega}\right)^k \binom{k}{k/2} & \text{if } k \text{ is even} \end{cases}$$
(3)

with $k = \sum_{j=1}^m k_j$.

B. Elements of differential geometry and complex analysis

We refer to [45], [46], [47] for comprehensive references on these topics. Let $x, x_0 \in \mathbb{R}^n$, $t \in \mathbb{R}_+$, and let the parameter ϵ vary in the range $(0, \epsilon_0]$ with $\epsilon_0 \ll 1$. Let $f, g : \mathbb{R}_+ \times \mathbb{R}^n \to \mathbb{R}^n$ be smooth time-varying vector fields. Define their Lie bracket according to

$$[g,f]=rac{\partial f(t,x)}{\partial x}g(t,x)-rac{\partial g(t,x)}{\partial x}f(t,x)\,.$$

In what follows, we will use the notation $\operatorname{ad}_g^0 f = f$, $\operatorname{ad}_g f = [g, f]$ and $\operatorname{ad}_g^k f = \operatorname{ad}_g^{k-1}[g, f]$. Given a diffeomorphism ϕ and a vector field f, the *pull-back of* f *along* ϕ , denoted by $\phi^* f$, is the vector field

$$(\phi^*f)(x)=\left(rac{\partial\phi^{-1}}{\partial x}\circ f\circ\phi
ight)(x).$$

A useful diffeomorphism is given by the flow map $\Phi_{0,T}^f$, that assigns to each point x_0 the value at time T of the solution of the initial value problem $\dot{x} = f(t, x)$, $x(0) = x_0$.

Given a positive scalar σ , define the complex σ -neighborhood of x_0 in \mathbb{C}^n as $B_{\sigma}(x_0) = \{z \in \mathbb{C}^n : ||z - x_0|| < \sigma\}$. Let f be a real analytic function on \mathbb{R}^n that admits a bounded analytic continuation over $B_{\sigma}(x_0)$. The norm of f is defined as

$$\|f\|_\sigma = \max_{z\in B_\sigma(x_0)} |f(z)|,$$

where f denotes both the function over \mathbb{R}^n and its analytic continuation. Given a timevarying vector field $(q,t) \mapsto Y(q,t) = Y_t(q)$, let Y_t^i be its *i*th component with respect to the usual basis on \mathbb{R}^n . Assuming $t \in [0, T]$, and assuming that every component function Y_t^i is analytic over $B_{\sigma}(x_0)$, we define the norm of Y as

$$\|Y\|_{\sigma,T} = \max_{t \in [0,T]} \max_{i \in \{1,...,n\}} \|Y^i_t\|_{\sigma}.$$

In what follows, we shall simplify notation by neglecting the subscript T in the norm of a time-varying vector field.

III. COORDINATE-FREE AVERAGING UNDER OSCILLATORY CONTROLS

We study averaging under oscillatory controls, using tools from the standard treatment on averaging (the first-order averaging theorem, see [17]), and from differential geometry (the variation of constants formula and the notion of pull-back vector field [47], [45]).

Let $x:[0,T]
ightarrow \mathbb{R}^n$ be the solution to the initial value problem

$$rac{dx}{dt} = f(t,x) + rac{1}{\epsilon}g\left(rac{t}{\epsilon},t,x
ight), \qquad x(0) = x_0 \ .$$

Enlarge the state space by considering x' = (t, x), denote by $\tau = t/\epsilon$ the fast time scale, and rewrite equation (4) as

$$rac{dx'}{d au} = \epsilon f'(x') + g'(au, x') \ , \quad x_0' = (0, x_0) \ , \qquad (5)$$

where the vector fields f' and g' are defined according to

$$f'(x') = (1, f(t, x)) \;, \qquad g'(au, x') = (0, g(au, t, x)) \;.$$

In the extended space, au is the independent variable and (t, x) are dependent variables. We write the flow of g' as

$$\Phi^{g'(au,x')}_{0, au}\left(t,x
ight)=\left(t,\Phi^{g(au,t,x)}_{0, au}(x)
ight)$$
 ,

and define the *pull-back vector field* F' as

$$F'(\tau, x') = \left(\left(\Phi_{0,\tau}^{g'(\tau, x')} \right)^* f' \right) (x').$$
(6)

Note that F' is of the form

$$F'(\tau, x') = (1, F(\tau, x')).$$
 (7)

Now, we give a novel sufficient condition to ensure that the pull-back vector field F' is *T*-periodic.

Proposition III.1: Assume that the vector fields in $\{g_{\tau}, \tau \in [0,T]\}$ are uniformly integrable, analytic admitting bounded analytical continuations over $B_{\sigma}(x_0), \sigma > 0$,

commutative, T-periodic and zero mean in their first argument, i.e., $g(\tau + T, t, x) = g(\tau, t, x)$, and $\int_0^T g(\tau, t, x) d\tau = 0$. Then, the flow $\Phi_{0,\tau}^{g'(\tau,x')}$ and the vector field F' are T-periodic.

Proof: The assumptions on the family of vector fields $\{g_{\tau}, \tau \in [0,T]\}$ are automatically verified by the family $\{g'_{\tau}, \tau \in [0,T]\}$. Let x'(t) be the solution to the initial value problem

$$\dot{x}'(au) = g'(au, x') \ , \quad x'(0) = x'_0 \ .$$
 (8)

Note that $x'(au)=\Phi_{0, au}^{g'(au,x')}(x'_0).$ Let X'(au)=x'(au+T). Then

$$\dot{X}'(au) = g'(au+T,X') = g'(au,X') \ , \quad X'(0) = x'(T) \ .$$

Consequently, $X'(\tau) = x'(\tau)$ iff x'(T) = x'(0). To prove the latter statement we use the Volterra series [45]. The flow of (8) is formally represented by the expansion

$$egin{aligned} x'(au) &\equiv \mathrm{Id}(x'_0) + \sum_{k=1}^{+\infty} \int_0^ au ds_1 \int_0^{s_1} ds_2 \dots \int_0^{s_{k-1}} ds_k (g'(s_k,x'_0) \circ \dots \circ g'(s_1,x'_0)) &= \ &x'_0 + \int_0^ au g'(s,x'_0) ds + \int_0^ au \int_0^{s_1} (g'(s_2,x'_0) \circ g'(s_1,x'_0)) ds_2 ds_1 + \dots \end{aligned}$$

where the vector fields g' are interpreted as derivations of $C^{\infty}(\mathbb{R}^{n+1})$. Given the above hypothesis, the convergence of this series is guaranteed by Proposition 2.1 in [45]. Now, using integration by parts and the fact that the vector fields commute, we have

$$egin{split} \int_0^ au igg(\int_0^{s_1} g'(s_2,x_0') ds_2 igg) \circ g'(s_1,x_0') ds_1 = \ & \left(\int_0^ au g'(s,x_0') ds
ight)^2 - \int_0^ au g'(s_1,x_0') \circ \left(\int_0^{s_1} g'(s_2,x_0') ds_2
ight) ds_1 \,, \end{split}$$

and hence

$$\int_0^{ au} \int_0^{s_1} (g'(s_2,x_0') \circ g'(s_1,x_0')) ds_2 ds_1 = rac{1}{2} \left(\int_0^{ au} g'(s,x_0') ds
ight)^2$$

By induction, one can show that

$$\int_0^\tau ds_1 \int_0^{s_1} ds_2 \dots \int_0^{s_{k-1}} ds_k (g'(s_k, x_0') \circ \dots \circ g'(s_1, x_0')) = \frac{1}{k!} \left(\int_0^\tau g'(s, x_0') ds \right)^k.$$
(10)

Since by hypothesis $g'(\tau, x')$ is zero-mean, we conclude from (9) that x'(T) = x'(0).

Given the result in the proposition, we define the averaged vector field $\overline{F'}$ as

$$\overline{F'}(x') = rac{1}{T}\int_0^T F'(au,x')d au$$

It is straightforward to see that $\overline{F'}(x') = (1, \overline{F}(x'))$. Finally, let $y, z : [0, T] \to \mathbb{R}^n$ be solutions to the initial value problems

$$rac{dz}{dt} = F\left(rac{t}{\epsilon}, t, z
ight), \quad z(0) = x_0, \qquad (11)$$

$$rac{dy}{dt}=\overline{F}(t,y), \quad y(0)=x_0.$$
 (12)

The following theorem extends Lemma 2.2 in [26] to general nonlinear control systems with two time scales and presents a refinement of the approximation result. It is the first of the two main analysis theorems.

Theorem III.2 (Coordinate-free averaging) Under the same hypothesis of Proposition III.1 we have,

(i) for $t \in \mathbb{R}_+$, we have

$$x(t)=\Phi^{g(au,t,x)}_{0,t/\epsilon}(z(t))$$
 ,

and, as $\epsilon
ightarrow 0$ on the time scale 1, $z(t) - y(t) = O(\epsilon),$

(ii) additionally, if f and g do not depend explicitly on the slow time scale t, i.e., f = f(x) and $g = g(t/\epsilon, x)$ and the origin is a hyperbolically stable critical point for $\overline{F} = \overline{F}(x)$, then $z(t) - y(t) = O(\epsilon)$ as $\epsilon \to 0$ for all $t \in \mathbb{R}_+$ and the differential equation (11) possesses a unique periodic orbit (which is hyperbolically stable) belonging to a $O(\epsilon)$ neighborhood of the origin,

(iii) if T = O(1), then $x(t) = \Phi_{0,(t/\epsilon \mod T)}^{g(\tau,t,x)}(y(t)) + O(\epsilon)$, on the time scale 1, where $\lfloor s \rfloor$ denotes the greatest integer less than or equal to $s \in \mathbb{R}$, and $(t/\epsilon \mod T)$ denotes $t/\epsilon - \lfloor t/(\epsilon T) \rfloor T$.

Proof: Recall the variation of constants formula (cf. [45]) to express the flow of the initial value problem $\dot{x} = f(t,x) + g(t,x)$, $x(0) = x_0$ at time T > 0,

$$x(T)=\Phi^g_{0,T}(z(T)), ext{ with } \dot{z}(t)=\left(\left(\Phi^g_{0,t}
ight)^*f
ight)(z), ext{ } z(0)=x_0.$$

Applying it to (5), we get

$$egin{aligned} &rac{dx'}{d au} = g'(au,x')\,, & x'(0) = z'(au)\,, \ &rac{dz'}{d au} = \epsilon F'(au,z')\,, & z'(0) = x_0'\,. \end{aligned}$$

Averaging this last system in z', we obtain

$$rac{dy'}{d au} = \epsilon \, \overline{F'}(y') \,, \qquad y'(0) = x'_{0} \,.$$

By the theorem of first order averaging (cf. [17], pages 39 and 71, and [48], page 168), we know that $z'(\tau) - y'(\tau) = O(\epsilon)$ over the time scale $\tau = 1/\epsilon$. Now, if we write y' = (v, y), we get from the previous equation that v = t and, changing the time scale back to $t = \epsilon \tau$,

$$rac{dy}{dt} = \overline{F}(t,y) \ , \qquad y(0) = x_{ extsf{0}} \ ,$$

which is the definition of equation (12). Putting z' = (u, z), we also deduce that u = tand $z(t) - y(t) = O(\epsilon)$ over the time scale 1. In addition, we recover equation (11)

$$rac{dz}{dt} = F\left(rac{t}{\epsilon},t,z
ight)\,, \qquad z(0) = x_{0}\,,$$

and from (13) we get $x(t) = \Phi^{g(au,t,x)}_{0,t/\epsilon}(z(t)).$

As for the second statement, in case f = f(x) and $g = g(t/\epsilon, x)$, if the origin is a hyperbolically stable critical point for \overline{F} , then the result follows from the theorem of first order averaging [17].

Finally, since g is T-periodic and zero-mean,

$$\int_0^{t/\epsilon} g(s,t,x) ds = \int_0^{(t/\epsilon \mod T)} g(s,t,x) ds + \int_{(t/\epsilon \mod T)}^{t/\epsilon} g(s,t,x) ds = \int_0^{(t/\epsilon \mod T)} g(s,t,x) ds,$$

and, using the Volterra series expansion for the flow $\Phi_{0,t/\epsilon}^{g(s,t,x)}$ (see [45]) and eq. (10),

$$egin{aligned} \Phi^{g(au,t,x)}_{0,t/\epsilon}(z(t)) &= z(t) + \sum_{k=1}^{+\infty} rac{1}{k!} \left(\int_{0}^{t/\epsilon} g(au,z(t)) ds
ight)^k \ &= z(t) + \sum_{k=1}^{+\infty} rac{1}{k!} \left(\int_{0}^{t/\epsilon} rac{\mathrm{mod}\,T}{g(au,z(t))} ds
ight)^k = \Phi^{g(au,t,x)}_{0,(t/\epsilon\,\,\mathrm{mod}\,T)}(z(t)) \,. \end{aligned}$$

Recall that $z(t) = y(t) + O(\epsilon)$ on the time scale 1. Since $(t/\epsilon \mod T) = O(1)$ and since the flow along g depends continuously on its initial condition we have

$$x(t)=\Phi^{g(au,t,x)}_{0,t/\epsilon}(z(t))=\Phi^{g(au,t,x)}_{0,(t/\epsilon \mod T)}(y(t)+O(\epsilon))=\Phi^{g(au,t,x)}_{0,(t/\epsilon \mod T)}(y(t))+O(\epsilon).$$

Now, we develop novel series expansions for the averaged system for multiple input systems of the form,

$$rac{dx}{dt} = f(x) + rac{1}{\epsilon} \sum_{i=1}^m u_i\left(rac{t}{\epsilon}, t
ight) g_i(x).$$
 (14)

Accordingly, we shall consider the (multinomial) iterated integrals $U_{(i_1,\ldots,i_k)}(\tau,t)$ and their averages $\overline{U}_{(i_1,\ldots,i_k)}(t)$ with respect to the first variable of the inputs $u_i(\tau,t)$. The following theorem is the second main analysis result.

Theorem III.3 (Multiple input system) Let $(\tau, t) \mapsto u_1(\tau, t), \ldots, u_m(\tau, t)$ be bounded functions, T-periodic and zero-mean in τ , continuously differentiable in t. Let g_1, \ldots, g_m be commuting vector fields. Then,

(i) the pull-back vector field F defined in eq. (7) satisfies

$$egin{aligned} F(au,t,x) &= f(t,x) + \sum_{k=1}^{+\infty} \sum_{(i_1,\ldots,i_k)\in\mathcal{I}} U_{(i_1,\ldots,i_k)}(au,t) \, \mathrm{ad}_{g_{i_1}}\ldots \mathrm{ad}_{g_{i_k}} \, f(t,x) \ &- \sum_{i=1}^m rac{\partial U_{(i)}}{\partial t}(au,t) g_i(x) \end{aligned}$$
 (15a)

$$=\sum_{k_{1},...,k_{m}=0}^{+\infty} \mathcal{U}_{k_{1},...,k_{m}}(\tau,t) \operatorname{ad}_{g_{1}}^{k_{1}} \ldots \operatorname{ad}_{g_{m}}^{k_{m}} f - \sum_{i=1}^{m} \frac{\partial U_{(i)}}{\partial t}(\tau,t) g_{i}(x),$$
(15b)

and its average \overline{F} satisfies

$$\overline{F}(t,x) = f(t,x) + \sum_{k=1}^{+\infty} \sum_{(i_1,...,i_k)\in\mathcal{I}} \overline{U}_{(i_1,...,i_k)}(t) \operatorname{ad}_{g_{i_1}} \dots \operatorname{ad}_{g_{i_k}} f(t,x) - \sum_{i=1}^m \frac{d\overline{U}_{(i)}}{dt}(t)g_i(x) \quad (16a)$$
$$= \sum_{k_1,...,k_m=0}^{+\infty} \overline{U}_{k_1,...,k_m}(t) \operatorname{ad}_{g_1}^{k_1} \dots \operatorname{ad}_{g_m}^{k_m} f - \sum_{i=1}^m \frac{d\overline{U}_{(i)}}{dt}(t)g_i(x). \quad (16b)$$

(ii) if f and g_1, \ldots, g_m , are analytic admitting bounded analytical continuations over $B_{\sigma}(x_0)$ and

$$T\sum_{j=1}^{m} \left\| u_{j} \right\|_{\mathcal{L}_{\infty}} \left\| g_{j} \right\|_{\sigma} < \frac{\sigma - \sigma'}{4n} \,. \tag{17}$$

for $0 < \sigma' < \sigma$, where $||u||_{\mathcal{L}_{\infty}}$ denotes the supremum of the absolute value of $u(\tau, t)$ for $\tau, t \in \mathbb{R}_+$, then the series expansions (15) and (16) converges absolutely and uniformly for $x \in B_{\sigma'}(x_0)$ and $t \in \mathbb{R}_+$.

Proof: We first prove the result for the single input case. Let us compute F' as in equation (6), where we let f' = f'(x') be τ -invariant and $g' = g'(\tau, x')$ be τ -varying. The following statement is proved in [47, Theorem 4.2.31]

$$rac{d}{d au}\left(\left(\Phi^{g'}_{\mathfrak{0}, au}
ight)^*f'
ight)(au,x')=\left(\Phi^{g'}_{\mathfrak{0}, au}
ight)^*[g'(au,x'),f'(x')].$$

At fixed $x' \in \mathbb{R}^{n+1}$, we integrate the previous equation from time 0 to au to obtain

$$\left(\left(\Phi^{g'}_{0, au}
ight)^*f'
ight)(au,x')=f'(x')+\int_0^ au(\Phi^{g'}_{0,s})^*[g'(s,x'),f'(x')]ds.$$

Iteratively applying the previous equality, we get

$$\left(\left(\Phi_{0, au}^{g'}
ight)^{*}f'
ight)(au,x')=f'(x')+\sum_{k=1}^{+\infty}\int_{0}^{ au}\ldots\int_{0}^{s_{k-1}}\left(\operatorname{ad}_{g'(s_{k},x')}\ldots\operatorname{ad}_{g'(s_{1},x')}f'(x')
ight)\,ds_{k}\ldots ds_{1}$$

Now, it can be proven by induction that

$$\operatorname{ad}_{g'(s_1,t,x)} f' = \left(0, u(s_1,t) \operatorname{ad}_{g(x)} f - rac{\partial u}{\partial t}(s_1,t)g(x)
ight) \,, \ \operatorname{ad}_{g'(s_k,x')} \ldots \operatorname{ad}_{g'(s_1,x')} f' = \left(0, u(s_k,t) \ldots u(s_1,t) \operatorname{ad}_{g(x)}^k f
ight) \,.$$

with $k \geq 2$. Finally,

$$F(au,t,x)=f(t,x)+\sum_{k=1}^{+\infty}U_k(au,t)\operatorname{ad}_g^kf(t,x)-rac{\partial U_1}{\partial t}(au,t)g(x),$$

and the result follows. In the multiple input case, the series expansions (15a) and (16a) can be deduced in the same way. As for (15b) and (16b), the fact that g_1, \ldots, g_m commute implies

$$\mathrm{ad}_{g_{i_{\sigma}(1)}}\ldots\mathrm{ad}_{g_{i_{\sigma}(k)}}\;f=\mathrm{ad}_{g_{i_{1}}}\ldots\mathrm{ad}_{g_{i_{k}}}\;f$$
 ,

for any $\sigma \in \Sigma_k$. Then, for each k,

$$\sum_{(i_1,\ldots,i_k)\in\mathcal{I}}U_{(i_1,\ldots,i_k)}(au,t)\operatorname{ad}_{g_{i_1}}\ldots\operatorname{ad}_{g_{i_k}}f(t,x)=\sum_{\substack{k_1,\ldots,k_m\geq 0\k_1+\cdots+k_m=k}}\mathcal{U}_{k_1,\ldots,k_m}(au,t)\operatorname{ad}_{g_1}^{k_1}\ldots\operatorname{ad}_{g_m}^{k_m}f(t,x)$$

As for the convergence of the series, from Lemma II.2 and Proposition 3.1 in [45],

$$egin{aligned} & \left\|\mathcal{U}_{k_1,\ldots,k_m}(au,t)\,\mathrm{ad}_{g_1}^{k_1}\ldots\,\mathrm{ad}_{g_m}^{k_m}\,f
ight\|_{\sigma'} &\leq \left(rac{4nT}{\sigma-\sigma'}
ight)^k \|u_1\|_{\mathcal{L}_\infty}^{k_1}\ldots\,\|u_m\|_{\mathcal{L}_\infty}^{k_m}\,\|g_1\|_{\sigma}^{k_1}\ldots\,\|g_m\|_{\sigma}^{k_m}\,\|f\|_{\sigma}\ &\leq \left(rac{4nT}{\sigma-\sigma'}
ight)^k\,\left(\sum_{j=1}^m \|u_j\|_{\mathcal{L}_\infty}\,\|g_j\|_{\sigma}
ight)^k\,\|f\|_{\sigma}\,\,, \end{aligned}$$

where $k = k_1 + \cdots + k_m$. As a consequence, equation (17) implies that the series in F is convergent. This also implies the convergence of the series expansions of \overline{F} .

Note that in the single-input case, m = 1, both series in (15) (resp. (16)) coincide.

IV. EXTENSIONS AND APPLICATIONS

In this section we investigate classes of differential equations for which the series expansions in Section III assume a particular structure. By doing so, we recover and extend a variety of earlier results on bilinear, polynomial and Hamiltonian systems.

Before proceeding, we summarize the averaging procedure from Theorem III.2 as

$$x(t)=\Phi^{g(t,x)}_{0,(t/\epsilon \mod T)}(y(t))+O(\epsilon)\,,\; \dot{y}=\overline{F}(y),\; y(0)=x_{0}\,.$$

For simplicity, we will pay special attention to single input systems with a single time scale, i.e., $g(\tau, t, x) = u(\tau)g(x)$.

A. Homogeneous systems

Here we focus on homogeneous systems. Let f be a vector field on \mathbb{R}^n . We say that f is homogeneous of degree i if each of its components with respect to the usual basis of \mathbb{R}^n is a homogeneous function of degree i. The set of homogeneous vector fields of degree i is denoted by \mathcal{H}_i . We have, for instance, that \mathcal{H}_0 is the set of constant vector fields and \mathcal{H}_1 is the set of linear vector fields. By convention, $\mathcal{H}_i = \{0\}$, for $i \leq -1$. If $f \in \mathcal{H}_i$ and $g \in \mathcal{H}_j$, then $[f,g] \in \mathcal{H}_{i+j-1}$.

It is straightforward to obtain the relevant quantities from Theorems III.2 and III.3



Fig. 1. Table of degrees of the drift vector field f and the input vector field g for homogeneous systems. The (i, j)th position refers to the case when $f \in \mathcal{H}_i$ and $g \in \mathcal{H}_j$.

for the lower triangular cases (0,0), (0,1) and (1,0) in Figure 1. Indeed, we have that

$$\begin{array}{ll} Case \ (0,0): \dot{x} = a + \frac{1}{\epsilon}u\left(\frac{t}{\epsilon}\right)b\,, \quad x(0) = x_0\,, \\ & \Phi_{0,(t/\epsilon \mod T)}^{u(t)b}(x_0) = b\int_0^{(t/\epsilon \mod T)} u(\tau)d\tau + x_0, \quad \overline{F} = a\,. \\ Case \ (0,1): \dot{x} = a + \frac{1}{\epsilon}u\left(\frac{t}{\epsilon}\right)Bx\,, \quad x(0) = x_0\,, \\ & \Phi_{0,(t/\epsilon \mod T)}^{u(t)Bx}(x_0) = e^{B\int_0^{(t/\epsilon \mod T)} u(\tau)d\tau}x_0\,, \quad \overline{F} = a + \sum_{k=1}^{+\infty}\overline{U}_k(-B)^k a \\ Case \ (1,0): \dot{x} = Ax + \frac{1}{\epsilon}u\left(\frac{t}{\epsilon}\right)b\,, \quad x(0) = x_0\,, \\ & \Phi_{0,(t/\epsilon \mod T)}^{u(t)b}(x_0) = b\int_0^{(t/\epsilon \mod T)} u(\tau)d\tau + x_0\,, \quad \overline{F} = Ax + \overline{U}_1Ab\,. \end{array}$$

In the following sections, we consider the cases of bilinear and polynomial systems.

B. Bilinear systems

We refer the reader to [14, Section 2.4] for a treatment on bilinear systems. Let

$$\dot{x} = Ax + rac{1}{\epsilon}u\left(rac{t}{\epsilon}
ight)Bx, \quad x(0) = x_0.$$
 (18)

Note that this system corresponds to the case (1, 1) in Figure 1. Lie brackets between linear vector fields are expressed in terms of matrix commutators

$$\operatorname{ad}_{Bx} Ax = (\operatorname{ad}_B A)x$$
, where $\operatorname{ad}_B A = AB - BA$

One can compute

$$\Phi^{gu(t)}_{0,(t/\epsilon \mod T)}(x_0) = \mathrm{e}^{B\int_0^{(t/\epsilon \mod T)} u(au)d au} x_0\,, \quad \overline{F}(x) = \left(A + \sum_{k=1}^{+\infty} \overline{U}_k \,\mathrm{ad}_B^k\,A
ight) x.$$

The following proposition shows a particular structure of the series expansion (16) and extends a result in [23].

Proposition IV.1: Consider the bilinear control system

$$\dot{x} = Ax + rac{1}{\epsilon}\sum_{i=1}^m u_i\left(rac{t}{\epsilon},t
ight)B_ix$$

and assume $B_i B_j = 0$ for all i, j. Consider also

$$\dot{y}=\left(A+\sum_{i=1}^m\overline{U}_{(i)}(t)\operatorname{ad}_{B_i}A+\sum_{i,j=1}^m\overline{U}_{(i,j)}(t)\operatorname{ad}_{B_i}\operatorname{ad}_{B_j}A-\sum_{i=1}^m\left(rac{d}{dt}\overline{U}_{(i)}(t)
ight)B_i
ight)y$$
 ,

with initial condition y(0) = x(0). Then,

$$x(t) = \mathrm{e}^{\sum_{i=1}^m B_i \int_0^{(t/\epsilon \mod T)} u_i(au,t) d au} y(t) + O(\epsilon)$$

The result follows from equation (16) by noting that $ad_{B_i} ad_{B_j} ad_{B_k} A = 0$ for all i, j, k.

C. Polynomial systems

Consider the system

$$\dot{x}=f(x)+rac{1}{\epsilon}u\left(rac{t}{\epsilon}
ight)g, \quad x(0)=x_{0},$$
 (19)

where the components of the vector field f are polynomials in x of degree at most M, and where the vector field g(x) = g is constant. This system is the combination of a finite number of (i, 0) cases in Figure 1. This structure leads to the following simplifications. The degree of $\operatorname{ad}_g^k f$ is M - k, and only the first M Lie brackets are non-vanishing. Accordingly, we have that

$$\Phi^{gu(t)}_{0,(t/\epsilon \mod T)}(x_0) = x_0 + \left(\int_0^{(t/\epsilon \mod T)} u(au) d au
ight) g \ , \ \overline{F}(x) = f(x) + \sum_{k=1}^M \overline{U}_k rac{\partial^k f}{\partial x^k}(\underbrace{g,\ldots,g}_{k ext{ times}})(x) \ .$$

Note that \overline{F} is a finite sum of polynomial vector fields.

D. Second order systems

We next focus on control systems described by second order differential equations. This setting is representative of interesting examples and very instructive. Consider the second order system on \mathbb{R}^n

$$\ddot{x} = rac{1}{\epsilon} u\left(rac{t}{\epsilon}
ight) g(x) \,.$$
 (20)

To write the equation in the standard (first order) form (14), define the vector fields on \mathbb{R}^{2n}

$$f(x,\dot{x}) = egin{bmatrix} \dot{x} \ 0 \end{bmatrix}, \quad g(x,\dot{x})^{ ext{lift}} = egin{bmatrix} 0 \ g(x) \end{bmatrix},$$

and compute the relevant Lie brackets as

$$\mathrm{ad}_{g^{\mathrm{lift}}} \, f = egin{bmatrix} g \ -rac{\partial g}{\partial x} \dot{x} \end{bmatrix}$$
 , $\mathrm{ad}_{g^{\mathrm{lift}}}^2 \, f = -\langle g : g
angle^{\mathrm{lift}}$, $\mathrm{ad}_{g^{\mathrm{lift}}}^k \, f = 0$,

for k>2 and where we define the operation of *symmetric product* between vector fields g_a, g_b on \mathbb{R}^n as

$$\langle g_a : g_b
angle = rac{\partial g_a}{\partial x} g_b + rac{\partial g_b}{\partial x} g_a \, .$$

From Theorems III.2 and III.3, we have

$$egin{aligned} \Phi^{g(x,\dot{x})^{ ext{lift}}u(t)}_{0,t}egin{pmatrix} x_0\ \dot{x}_0\end{pmatrix} &=egin{pmatrix} x_0\ \dot{x}_0\end{pmatrix} + egin{pmatrix} 0\ igl(\int_0^t u(au)d auigr)g(x_0)\end{bmatrix},\ \overline{F} &= f + \overline{U}_1 ext{ ad}_{g^{ ext{lift}}}f + \overline{U}_2 ext{ ad}_{g^{ ext{lift}}}^2 f = egin{pmatrix} \dot{x}\ 0\end{bmatrix} + \overline{U}_1egin{pmatrix} g\ - egin{pmatrix} g\ - eta_2\ igl(g:g)\end{bmatrix},\ \overline{F} &= f + \overline{U}_1 ext{ ad}_{g^{ ext{lift}}}f = egin{pmatrix} \dot{x}\ 0\end{bmatrix} + \overline{U}_1egin{pmatrix} g\ - eta_2\ igl(g:g)\end{bmatrix}, \end{aligned}$$

so that, using the variables (y_1, y_2) for the averaged system, we write

$$\dot{y}_1=y_2+\overline{U}_1g(y_1)\,,\quad \dot{y}_2=-\overline{U}_1rac{\partial g}{\partial x}(y_1)y_2-\overline{U}_2\langle g:g
angle(y_1)$$

with initial conditions $(y_1(0), y_2(0)) = (x(0), \dot{x}(0))$. It is instructive to compute the second time derivative of y_1 , and write the averaged system again as an equation of second order. Some straightforward simplifications lead to

$$\ddot{y}_1 = \left(rac{1}{2}\overline{U}_1^2 - \overline{U}_2
ight) \langle g:g
angle(y_1),$$
 (21)

with initial conditions $(y_1(0),\dot{y}_1(0))=(x(0),\dot{x}(0)+\overline{U}_1g(x(0))).$ In summary, we have

$$x(t)=y_1(t)+O(\epsilon)\,,\quad \dot{x}(t)=\dot{y}_1(t)+g(y_1(t))\left(\int_0^{(t/\epsilon \mod T)}u(s)ds-\overline{U}_1
ight)+O(\epsilon)\,.$$

Remark IV.2: Analogues to the result in equation (21) and their physical meaning have been long studied; e.g., see [21], [22], [24], [25], [26]. In particular, if g is a potential

field, $g = \partial V/\partial x$, then one can compute $\langle g : g \rangle = \partial W/\partial x$, where $W = (\partial V/\partial x)^2$ is the classical Kapitsa's potential [21], [24] (also called the averaged potential). It is easy to see that every isolated critical point of V is a minimum of W. Using Hölder inequality, we obtain

$$egin{aligned} &rac{1}{2}\overline{U}_1^2-\overline{U}_2 = rac{1}{2T}\left(rac{1}{T}\left(\int_0^T U_1(s)ds
ight)^2 - \int_0^T U_1^2(s)ds
ight)\ &< rac{1}{2T}\left(rac{1}{T}\left(\left(\int_0^T U_1^2(s)ds
ight)^rac{1}{2}\left(\int_0^T 1ds
ight)^rac{1}{2}
ight)^2 - \int_0^T U_1^2(s)ds
ight) = 0\,, \end{aligned}$$

and hence every isolated equilibrium point of the original system is a Lyapunov-stable equilibrium point for the averaged system [25].

Reasoning as before, one can prove the following more general result.

Proposition IV.3: Consider the control system

$$\ddot{x}+f_1(x)\dot{x}+f_0(x)=rac{1}{\epsilon}\sum_i u_i\left(rac{t}{\epsilon},t
ight)\,g_i(x)$$
 ,

and the initial value problem

$$\ddot{y}+f_1(y)\dot{y}+f_0(y)=rac{1}{2}\sum_{i,j}\Big(\overline{U}_{(i)}(t)\overline{U}_{(j)}(t)-\overline{U}_{(i,j)}(t)-\overline{U}_{(j,i)}(t)\Big)\langle g_i:g_j
angle(y)$$

with initial conditions $y(0)=x(0),\,\dot{y}(0)=\dot{x}(0)+\sum_i\overline{U}_{(i)}(0)g_i(x(0)).$ Then, we have

E. Hamiltonian control systems

Second order systems as in equation (20) are examples of Lagrangian control systems, and the analysis presented above can be generalized to a coordinate-free setting on manifolds; e.g., the result in equation (21) agrees with the results in [26]. We present here a coordinate-free based treatment for Hamiltonian control systems as described for example in [49], [13]. Consider the control system

$$\dot{x} = X_H + \sum\limits_{i=1}^m rac{1}{\epsilon} u_i\left(rac{t}{\epsilon}
ight) X_{H_i} \,, \; x(0) = x_{0} \,,$$

where $x = (q, p) \in T^* \mathbb{R}^n$, H, $H_i \in C^{\infty}(T^* \mathbb{R}^n)$, and X_H , X_{H_i} denote the corresponding Hamiltonian vector fields with respect to the canonical symplectic form $\Omega_{\mathbb{R}^n}$ on $T^* \mathbb{R}^n$. Let $\{\cdot, \cdot\}$ denote the Poisson bracket associated with $\Omega_{\mathbb{R}^n}$. Assume that the vector fields X_{H_1}, \ldots, X_{H_m} commute, or equivalently, that the Poisson bracket $\{H_i, H_j\}$ is constant for any $i, j \in \{1, \ldots, m\}$.

The pull-back vector field F in Theorem III.3 is again Hamiltonian with respect to

$$H^* = H + \sum_{k=1}^{+\infty} (-1)^k \sum_{(i_1,...,i_k) \in I} U_{(i_1,...,i_k)} \{H_{i_1}, \{H_{i_2}, \ldots, \{H_{i_k}, H\}\} \dots \} \,.$$

In particular, let $\varphi_1, \ldots, \varphi_m$ be functions defined on \mathbb{R}^n and consider their natural lift, $\varphi_i^{\text{lift}} = \varphi_i \circ \pi_{\mathbb{R}^n}$, where $\pi_{\mathbb{R}^n} : T^*\mathbb{R}^n \to \mathbb{R}^n$ is the canonical projection. Then, it is straightforward to verify that $\{\varphi_i^{\text{lift}}, \varphi_j^{\text{lift}}\} = 0$, and we can consider a control system with input Hamiltonian functions $H_i = \varphi_i^{\text{lift}}$. If the Hamiltonian H has a polynomial dependence on the momentum variables p, say of order l, then one can see that the series for H^* is finite,

$$H^* = H + \sum_{k=1}^l (-1)^k \sum_{(i_1,...,i_k) \in I} U_{(i_1,...,i_k)} \{ arphi_{i_1}^{ ext{lift}}, \{ arphi_{i_2}^{ ext{lift}}, \ldots, \{ arphi_{i_k}^{ ext{lift}}, H \} \} \dots \}$$

This is the case, for instance, of the so-called simple mechanical systems, where the Hamiltonian corresponds to kinetic plus potential energy, $H = \frac{1}{2}p^T M^{-1}(q)p + V(q)$, with M the mass matrix. Indeed, one gets $\{\varphi_i^{\text{lift}}, \{\varphi_j^{\text{lift}}, \{\varphi_k^{\text{lift}}, H\}\} = 0$.

F. Systems with recurrence relations

Next, we investigate a summing method based on recursive Lie bracket relationships and generating functions [50]. Let $\langle g | f \rangle$ be the smallest *g*-invariant distribution containing *f*. Let the distribution $\langle g | f \rangle$ be finite-dimensional. Note that any pair of linear vector fields satisfies this assumption because of the Cayley-Hamilton theorem.

Lemma IV.4: Assume $\operatorname{ad}_g^p f = \lambda \operatorname{ad}_g^q f$ for some integers $p > q \ge 0$, where $\lambda : \mathbb{R}^n \to \mathbb{R}$ belongs to ker g. Then

$$\overline{F}(x) = \sum_{k=0}^{q-1} \overline{U}_k \operatorname{ad}_g^k f + \left(\sum_{k=0}^{+\infty} \overline{U}_{q+(p-q)k} \lambda^k
ight) \operatorname{ad}_g^q f + \dots + \left(\sum_{k=0}^{+\infty} \overline{U}_{(p-1)+(p-q)k} \lambda^k
ight) \operatorname{ad}_g^{p-1} f.$$

Let $_{(m)}H_{(n)}$ be the generalized hypergeometric function [44] of indexes $m, n \in \mathbb{N}$. If $u(t) = a \sin \omega t$, the generating function is

$$\sum_{k=0}^{+\infty} \overline{U}_{mk+n} \lambda^{k} = \frac{1}{\pi n!} \beta(\frac{1}{2}, \frac{1}{2} + n) \left(\frac{2a}{\omega}\right)^{n} {}_{(m+1)} H_{(2m)} \left(\left[\frac{1, \frac{1}{2m} + \frac{n}{m}, \frac{3}{2m} + \frac{n}{m}, \dots, \frac{2m-1}{2m} + \frac{n}{m}}{\frac{1+n}{m}, \frac{1+n}{m}, \frac{2+n}{m}, \frac{2+n}{m}, \dots, \frac{m+n}{m}, \frac{m+n}{m}} \right], \lambda \left(\frac{2a}{\omega m} \right)^{m} \right)$$

If $u(t) = a \cos \omega t$, the generating function for m even is

$$\sum_{k=0}^{+\infty} \overline{U}_{mk+n} \lambda^k = 0,$$

$$\sum_{k=0}^{+\infty} \overline{U}_{mk+n} \lambda^k = \frac{1}{\Gamma\left(1+\frac{n}{2}\right)^2} \left(\frac{a}{2\omega}\right)^n {}_{(1)}H_{(m)}\left(\left[\frac{1}{\frac{2+n}{m}, \frac{2+n}{m}, \frac{4+n}{m}, \frac{4+n}{m}, \dots, \frac{m+n}{m}, \frac{m+n}{m}\right], \lambda\left(\frac{a}{m\omega}\right)^m\right),$$

for n odd and even respectively; and for m odd we compute

$$\sum_{k=0}^{+\infty} \overline{U}_{mk+n} \lambda^{k} = \frac{1}{\Gamma\left(1 + \frac{m+n}{2}\right)^{2}} \lambda\left(\frac{a}{2\omega}\right)^{m+n}{}_{(1)} H_{(2m)} \left(\begin{bmatrix} 1\\ \frac{2+m+n}{2m}, \frac{2+m+n}{2m}, \dots, \frac{3m+n}{2m}, \frac{3m+n}{2m} \end{bmatrix}, \lambda^{2} \left(\frac{a}{2m\omega}\right)^{2m} \right)$$

$$\sum_{k=0}^{+\infty} \overline{U}_{mk+n} \lambda^{k} = \frac{1}{\Gamma\left(1 + \frac{n}{2}\right)^{2}} \left(\frac{a}{2\omega}\right)^{n}{}_{(1)} H_{(2m)} \left(\begin{bmatrix} 1\\ \frac{2+n}{2m}, \frac{2+n}{2m}, \frac{4+n}{2m}, \frac{4+n}{2m}, \dots, \frac{2m+n}{2m}, \frac{2m+n}{2m} \end{bmatrix}, \lambda^{2} \left(\frac{a}{2m\omega}\right)^{2m} \right),$$

for n odd and even respectively.

The simplest example of Lemma IV.4 is (p, q) = (1, 0). Accordingly,

$$\overline{F}_{\sin}(x)=f(x)\,\left({
m e}^{\left(rac{a\,\lambda}{\omega}
ight)}I_0\left(rac{a\,\lambda}{\omega}
ight)
ight),\quad \overline{F}_{\cos}(x)=f(x)\,I_0\left(rac{a\,\lambda}{\omega}
ight),$$

where I_0 denotes the modified Bessel function of the first kind. Another case which we will use later is (p,q) = (4,2),

$$egin{aligned} \overline{F}_{\sin}(x) &= f(x) + rac{a}{\omega} \operatorname{ad}_g f + rac{1}{\lambda} \operatorname{ad}_g^2 f \left(-1 + I_0 \left(rac{a \sqrt{\lambda}}{\omega}
ight) \cosh \left(rac{a \sqrt{\lambda}}{\omega}
ight)
ight) \ &+ rac{1}{\lambda \sqrt{\lambda}} \operatorname{ad}_g^3 f \left(-rac{a}{\omega} \sqrt{\lambda} + I_0 \left(rac{a \sqrt{\lambda}}{\omega}
ight) \sinh \left(rac{a \sqrt{\lambda}}{\omega}
ight)
ight) \ , \ \overline{F}_{\cos}(x) &= f(x) + rac{1}{\lambda} \operatorname{ad}_g^2 f \left(-1 + I_0 \left(rac{a \sqrt{\lambda}}{\omega}
ight)
ight) \ . \end{aligned}$$

V. ON STABILIZATION VIA OSCILLATORY CONTROLS

In this section we discuss the problem of stabilization of the nonlinear system $\dot{x} = f(x)$ by means of highly oscillatory controls of the form $(1/\epsilon)u(t/\epsilon)g(x)$. The starting point is the result in Theorem III.2 about the existence of hyperbolically stable periodic orbits for $\dot{x} = f(x)$ provided the origin is a hyperbolically stable equilibrium for \overline{F} . In some cases, we shall prove asymptotic stability for the original equilibrium point (this is what was termed as t-stabilizability in [38]) and in some others we shall prove that the equilibrium bifurcates to an asymptotically stable periodic orbit contained in an $O(\epsilon)$ -neighbor (v-stabilizability in [38]). We start by studying whether the origin is an equilibrium point for the averaged system. We say that a vector field $h : \mathbb{R}^n \to T\mathbb{R}^n$, $h(x) = (x, \tilde{h}(x))$ is odd (resp. even) iff $\tilde{h}(x) = -\tilde{h}(-x)$ (resp. $\tilde{h}(x) = \tilde{h}(-x)$).

Lemma V.1 (Equilibrium points) The origin is an equilibrium point of the averaged system, $\overline{F}(0) = 0$, if either of the following conditions are satisfied:

(i)
$$f(0) = g_1(0) = \cdots = g_m(0) = 0$$
,

(ii) f(0) = 0 and f is odd, g_j is even for all $1 \le j \le m$, there exists i such that $g_i(0) \ne 0$, and the odd multinomial iterated averages of the inputs vanish, i.e., $\overline{\mathcal{U}}_{k_1,\ldots,k_m} = 0$ whenever $k_1 + \cdots + k_m$ is odd.

Proof: In case (i), one can prove recursively that

$$\operatorname{ad}_{g_1}^{k_1}\ldots\operatorname{ad}_{g_m}^{k_m}f(0)=0\;,\quad k_1,\ldots,k_m\geq 0\;,$$

and hence $\overline{F}(0) = 0$. To see *(ii)*, consider the (k_1, \ldots, k_m) -th term $\overline{\mathcal{U}}_{k_1,\ldots,k_m} \operatorname{ad}_{g_1}^{k_1} \ldots \operatorname{ad}_{g_m}^{k_m} f$ in the expansion of \overline{F} . If f is odd and g_j is even for all j, the vector field $\operatorname{ad}_{g_1}^{k_1} \ldots \operatorname{ad}_{g_m}^{k_m} f$ is odd whenever $k_1 + \cdots + k_m$ is even. Accordingly, each term $\overline{\mathcal{U}}_{k_1,\ldots,k_m} \operatorname{ad}_{g_1}^{k_1} \ldots \operatorname{ad}_{g_m}^{k_m} f$ is either odd or it vanishes. Therefore, \overline{F} is an odd function, and $\overline{F}(0) = 0$.

Next, we study the linearization of the averaged system.

Proposition V.2: Assume $f(0) = g_1(0) = \cdots = g_m(0) = 0$. At the origin, the linearization of the averaged system equals the average of the linearized system.

Proof: We prove it for the single input setting. Let $f = \sum_{i=1}^{+\infty} f^{[i]}$, $g = \sum_{i=1}^{+\infty} g^{[i]}$ be the Taylor expansions around x = 0 of f and g. Accordingly, $f^{[i]}, g^{[i]} \in \mathcal{H}_i$ and

$$\mathrm{ad}_g^k \, f = \sum_{\substack{j=1 \ i_1, ..., i_k = 1}}^{+\infty} \mathrm{ad}_{g^{[i_1]}} \ldots \mathrm{ad}_{g^{[i_k]}} \, f^{[j]} = \mathrm{ad}_{g^{[1]}}^k \, f^{[1]} + h$$
 ,

where h is an infinite sum of homogeneous polynomials of degree ≥ 2 . Consequently,

$$rac{\partial}{\partial x}\left(\operatorname{ad}_{g}^{k}f
ight)\left(0
ight)=\operatorname{ad}_{rac{\partial g}{\partial x}\left(0
ight)}^{k}rac{\partial f}{\partial x}(0)$$
 ,

where one adjoint operator is a Lie bracket and the other a matrix commutator. This implies that the linearization of the averaged system is equal to

$$rac{\partial \overline{F}}{\partial x}(0) = rac{\partial f}{\partial x}(0) + \sum_{k=1}^{+\infty} \overline{U}_k \operatorname{ad}_{rac{\partial g}{\partial x}(0)}^k rac{\partial f}{\partial x}(0) \, ,$$

which is the average of the linearized system (see Section IV-B on bilinear systems).

Note that the setting of bilinear systems is very important as it represents the linearization of the average of any nonlinear system with $f(0) = g_1(0) = \cdots = g_m(0) = 0$.

Corollary V.3: Let $f(0) = g_1(0) = \cdots = g_m(0) = 0$. If the trace of the linearization of the drift vector field f is positive, then the averaged system is unstable for any oscillatory control law.

Proof: Since $tr(ad_C D) = 0$ for any matrix C, D, we have that

$$\mathrm{tr}\left(rac{\partial\overline{F}}{\partial x}(0)
ight) = \mathrm{tr}\left(\sum_{k_1,...,k_m\geq 0}^{+\infty}\overline{\mathcal{U}}_{k_1,...,k_m}\,\mathrm{ad}_{rac{\partial g_1}{\partial x}(0)}^{k_1}\ldots\mathrm{ad}_{rac{\partial g_m}{\partial x}(0)}^{k_m}rac{\partial f}{\partial x}(0)
ight) = \mathrm{tr}\left(rac{\partial f}{\partial x}(0)
ight) > 0\,,$$

and therefore the averaged system is unstable.

The corollary is a generalization in two directions of the result in [38] about the stabilizability of the system (18) by linear multiplicative vibrations. First, we do not require $\frac{\partial f}{\partial x}(0)$ to be nonderogatory. Recall that a matrix is nonderogatory if its eigenvalues have geometric multiplicity equal to one [51]. Second, we consider general nonlinear systems and vibrations. Next, we present a classical result on stabilization by means of oscillatory controls.

Proposition V.4: Consider the nonlinear system $\dot{x} = f(x)$, with f(0) = 0. If $A = \partial f/\partial x(0)$ is nonderogatory and tr A < 0, then there exist commuting linear vector fields $\{g_1, \ldots, g_{n-1}\}$ and oscillatory controls $\{u_1, \ldots, u_{n-1}\}$ such that x = 0 is locally asymptotically stable for

$$\dot{x}=f(x)+rac{1}{\epsilon}\sum\limits_{i=1}^{n-1}u_i\left(rac{t}{\epsilon}
ight)g_i(x)\,.$$
 (22)

Proof: The proof goes along the lines of [37], [38]. Assume A is in companion form (otherwise, we first perform a change of coordinates). This means that

$$A = egin{pmatrix} 0 & 1 & 0 & \dots & 0 \ 0 & 0 & 1 & \dots & 0 \ & & dots & & & \ 0 & \dots & \dots & 0 & 1 \ a_n & a_{n-1} & \dots & a_2 & a_1 \end{pmatrix} = ext{Comp}(a_n,\dots,a_2,a_1) \,.$$

Let $i \in \{1, \ldots, n-1\}$ and consider the linear vector fields

$$g_i(x) = E_{ni}x$$

where E_{ni} is the matrix which has a 1 in the row n and column i and zero otherwise. Since $E_{ni}E_{nj} = 0$, for all $i, j \in \{1, ..., n - 1\}$, the input vector fields commute and the result in Proposition IV.1 applies. Furthermore, the only non-vanishing second order commutators are

$$\operatorname{ad}_{E_{n(n-1)}}^{2}A = -2E_{n(n-1)}, \quad \operatorname{ad}_{E_{nj}}\operatorname{ad}_{E_{n(n-1)}}A = -E_{nj},$$
 (23)

for j < n-1 and therefore the linearized averaged system is again in companion form. Taking the input functions $u_i = r_i \cos(\omega t)$, the first averages $\overline{U}_{(i)}$ vanish, and from Proposition IV.1 we have that $\overline{A} = \frac{\partial \overline{F}}{\partial x}(0)$ equals

$$egin{aligned} &A-2\,\overline{U}_{(n-1,n-1)}E_{n(n-1)}-\sum_{i=1}^{n-2}(\overline{U}_{(i,n-1)}+\overline{U}_{(n-1,i)})E_{ni}\ &= ext{Comp}\left(a_n-rac{r_1r_{n-1}}{2\omega^2},\,\ldots,a_2-rac{r_{n-1}^2}{2\omega^2},\,a_1
ight)\,, \end{aligned}$$

where in the last equality we have used the result in Lemma II.1 and the equality (3). By assumption $tr(A) = a_1 < 0$. Let λ_i denote the *i*th eigenvalue of A, define

$$ar{\lambda}_i = rac{a_1}{n} + j \mathrm{Im} \lambda_i$$
 ,

and the Hurwitz polynomial $(x - \overline{\lambda}_1) \dots (x - \overline{\lambda}_1) = x^n + \overline{a}_1 x^{n-1} + \dots + \overline{a}_{n-1} x + \overline{a}_n$. Note that $\overline{a}_1 = a_1$ and $\overline{a}_2 \leq a_2$. Now, it is clear that there exists an appropriate selection of the amplitudes r_i that makes $\overline{A} = \text{Comp}(\overline{a}_n, \dots, \overline{a}_2, a_1)$. Therefore y = 0 is locally asymptotically stable for the averaged system. From Theorem III.2, we know that equation $\dot{z}(t) = F(t/\epsilon, z)$ possesses a unique (asymptotically stable) periodic orbit $z_p(t)$ in a $O(\epsilon)$ -neighbourhood of the origin. Since z = 0 is trivially periodic, we deduce that $z_p(t) = 0$, and finally $x = \Phi_{0,t/\epsilon}^{g(t,x)}(0) = 0$ is a locally asymptotically stable equilibrium point for equation (22).

An interesting observation is that systems with positive trace may be stabilized by means of vibrations g with $g(0) \neq 0$: the example in [52] is a linear system $\dot{x} = Ax$,

with tr(A) > 0 and a control input $g = g^{[0]} + g^{[2]}$, $g^{[0]} \in \mathcal{H}_0$, $g^{[2]} \in \mathcal{H}_2$. Here, we give the following result.

Proposition V.5: Consider the linear system $\dot{x} = Ax$, and let A be a nonderogatory matrix with tr A > 0. Then, there exist a nonlinear vector field g_{n1} , commuting vector fields $\{g_1, \ldots, g_{n-1}\}$ and controls $\{u_1, \ldots, u_{n-1}\}$ such that the equilibrium x = 0 of the linear system becomes an asymptotically stable periodic orbit contained in an $O(\epsilon)$ neighborhood of the origin for the equation

$$\dot{x} = Ax + g_{ ext{nl}}(x) + rac{1}{\epsilon}\sum_{i=1}^{n-1} u_i\left(rac{t}{\epsilon}
ight)g_i(x)\,.$$

Proof: Assume A is in companion form. Let

$$g_{ ext{nl}}(x) = f^{[3]}(x) = (0, \dots, 0, bx_1^3 + cx_1^2 x_n) \in \mathcal{H}_3.$$

For $i \in \{2, \ldots, n-1\}$, define the commuting vector fields

$$g_1(x)=(1,0,\ldots,0)\in\mathcal{H}_0\;,\quad g_i(x)=E_{n\,i}x\in\mathcal{H}_1\,,$$

and the controls $u_i = r_i \cos(\omega t)$. The averaged system is

$$\overline{F}(x) = \sum_{k_1,\ldots,k_m=0}^{+\infty} \overline{\mathcal{U}}_{k_1,\ldots,k_m} \operatorname{ad}_{g_1}^{k_1} \ldots \operatorname{ad}_{g_m}^{k_m} (Ax + f^{[3]})$$

We first verify that the origin is an equilibrium point for the averaged system. Because of homogeneity arguments, the 0th order term in \overline{F} is

$$\overline{F}(0) = \sum_{k_2,\ldots,k_m=0}^{+\infty} \left(\overline{\mathcal{U}}_{1,k_2,\ldots,k_m} \operatorname{ad}_{g_2}^{k_2} \ldots \operatorname{ad}_{g_m}^{k_m} \left(\operatorname{ad}_{g_1} Ax\right) + \overline{\mathcal{U}}_{3,k_2,\ldots,k_m} \operatorname{ad}_{g_2}^{k_2} \ldots \operatorname{ad}_{g_m}^{k_m} \left(\operatorname{ad}_{g_1}^3 f^{[3]}
ight)
ight)_{x=0} \ .$$

Since $\operatorname{ad}_{g_i}\operatorname{ad}_{g_1}Ax=0$ and $\operatorname{ad}_{g_i}\operatorname{ad}_{g_1}^3f^{[3]}(x)=0$ for all $i,j\in\{2,\ldots,n-1\},$

$$\overline{F}(0)=\overline{\mathcal{U}}_{1,0,...,0}\,\mathrm{ad}_{g_1}\,Ax+\overline{\mathcal{U}}_{3,0,...,0}\,\mathrm{ad}_{g_1}\,f^{[3]}=0$$
 ,

where in the last equality we have used equation (3). With regards to the linearization of \overline{F} , we have that

$$\overline{A} = rac{\partial \overline{F}}{\partial x}(0) = \sum_{k_2,...,k_m=0}^{+\infty} \left(\overline{\mathcal{U}}_{0,k_2,...,k_m} \operatorname{ad}_{g_2}^{k_2} \ldots \operatorname{ad}_{g_m}^{k_m} Ax \ + \overline{\mathcal{U}}_{2,k_2,...,k_m} \operatorname{ad}_{g_2}^{k_2} \ldots \operatorname{ad}_{g_m}^{k_m} \left(\operatorname{ad}_{g_1} f^{[3]}
ight)
ight)_{x=0}.$$

The first term is computed as in the previous proposition via equation (23). As for the second one, note that for $i, j \in \{2, ..., n-1\}$,

$$\mathrm{ad}_{g_1}^2 \, f^{[3]} = (0, \dots, 6bx_1 + 2cx_n) \,, \quad \mathrm{ad}_{g_i} \, \mathrm{ad}_{g_1}^2 \, f^{[3]} = (0, \dots, 2cx_i) \,, \quad \mathrm{ad}_{g_i} \, \mathrm{ad}_{g_j} \, f^{[3]} = 0 \,.$$

Using again equation (3), we conclude that the linearized averaged system is also in companion form. Indeed, \overline{A} equals

$$ext{Comp}igg(a_n+rac{3br_1^2}{2\omega^2},a_{n-1}-rac{r_2r_{n-1}}{2\omega^2},\,\ldots,a_2-rac{r_{n-1}^2}{2\omega^2},\,a_1+rac{cr_1^2}{2\omega^2}igg)$$

Selecting $c < -a_1\omega^2/r_1^2$, we have that $\operatorname{tr} \overline{A} < 0$. Finally, an appropriate selection of the coefficients b, r_1, \ldots, r_{n-1} makes the linearization of the averaged system a Hurwitz matrix. As before, this implies that y = 0 is asymptotically stable for the averaged system. The application of Theorem III.2 concludes the proof.

Remark V.6: In summary, stabilization has been achieved building on strong nilpotency assumptions of the type $B_iB_j = 0$. In the following, we show that the less restrictive nilpotency properties discussed in Section IV-F may also be instrumental to provide alternative stabilization schemes. The interesting observation in this case is that the *full* series expansion is taken into account, not just a truncated version. Consider the bilinear system $\dot{x} = Ax + (1/\epsilon)u(t/\epsilon)Bx$ with $x \in \mathbb{R}^2$. Assume A is diagonal and tr(A) < 0. Write $A = A_1 + A_2$, where

$$A_1=\left(egin{array}{cc} {
m tr}(A)/2 & 0 \ 0 & {
m tr}(A)/2 \end{array}
ight) \,, \quad A_2=\left(egin{array}{cc} \epsilon & 0 \ 0 & -\epsilon \end{array}
ight) \,.$$

Note that A_1 is a stable matrix. Let $B = \begin{pmatrix} 0 & 1/2 \\ -1/2 & 0 \end{pmatrix}$, and compute $\operatorname{ad}_B^2 A = -A_2$, $\operatorname{ad}_B^4 A = A_2 = -\operatorname{ad}_B^2 A$.

Let $u = a \cos \omega t$ and, following Lemma IV.4 with (p, q) = (4, 2), compute

$$\overline{F}(x) = (A_1+A_2)x - \mathrm{ad}_B^2 A\left(-1 + I_0\left(rac{a}{\omega}\sqrt{-1}
ight)
ight)x = \left(A_1+A_2 I_0\left(rac{a}{\omega}\sqrt{-1}
ight)
ight)x$$
 .

The modified Bessel function with imaginary arguments possesses an infinite number of positive zeros. The smallest $z_1 \in \mathbb{R}_+$ such that $I_0(z_1\sqrt{-1}) = 0$ is an irrational number belonging to the interval [2.424, 2.425]. Therefore, the averaged system is asymptotically stable provided the input parameters satisfy $a = z_1 \omega$.

VI. ON TRACKING VIA OSCILLATORY CONTROLS

We present an application of our averaging analysis to the problem of trajectory tracking via oscillatory controls. We consider the tracking problem for second order systems, and we design control laws inspired by the inversion algorithm in [53]. In what follows, let i, j, k take values in $\{1, \ldots, m\}$ unless otherwise stated.

Consider the control system

$$\ddot{x} + f_1(x)\dot{x} + f_0(x) = \sum_i w_i \, g_i(x) \, ,$$
 (24)

and the following tracking problem: given a smooth desired curve $x^d : [0, T] \to \mathbb{R}^n$ with initial conditions $x^d(0) = x(0), \dot{x}^d(0) = \dot{x}(0)$, find controls laws $w_i : \mathbb{R}^{2n} \times [0, T] \to \mathbb{R}^m$ such that the solution x to equation (24) approximates x^d up to an error of order ϵ .

We make the following controllability assumption:

(A) The distribution span $\{g_i, \langle g_j : g_k \rangle\}$ is full rank, and $\langle g_j : g_j \rangle$ belongs to span $\{g_i\}$. Accordingly,

(i) there exist functions $z_i^d, z_{jk}^d: [0,T] o \mathbb{R}$, for j < k, such that

$$\ddot{x}^d+f_1(x^d)\dot{x}^d+f_0(x^d)=\sum_i z_i^d g_i(x^d)+\sum_{j< k} z_{jk}^d\langle g_j:g_k
angle(x^d)\,,$$

(ii) there exist smooth functions $\alpha_{ij}:\mathbb{R}^n
ightarrow\mathbb{R}$ such that

$$\langle g_i:g_i
angle(x)=\sum_jlpha_{ij}(x)g_j(x)\,,\qquad orall x\in\mathbb{R}^n$$

There are N = m(m-1)/2 pairs of integers (j,k), with j < k. Let $(j,k) \mapsto a(j,k) \in \{1,\ldots,N\}$ be a enumeration of these pairs, and define the scalar functions

$$\psi_{a(j,k)}(t)=\sqrt{2}\,a(j,k)\cos(a(j,k)t)$$
 .

Proposition VI.1: Let x^d : $[0,T] \to \mathbb{R}^n$ be a desired curve with initial conditions $x^d(0) = x(0), \dot{x}^d(0) = \dot{x}(0)$. The solution x to equation (24) equals x^d up to an error of

order ϵ over the time scale 1 when the control laws w_i are

$$egin{aligned} &w_i = v_i(t,x) + rac{1}{\epsilon} u_i\left(rac{t}{\epsilon},t
ight) \ , \ &v_i(t,x) = z_i^d(t) + rac{1}{2}\sum_j lpha_{ji}(x) \left(j-1 + \sum_{\ell=j+1}^m (z_{j\ell}^d(t))^2
ight) \ , \ &u_i(au,t) = -\sum_{\ell=1}^{i-1} \psi_{a(\ell,i)}(au) + \sum_{\ell=i+1}^m z_{i\ell}^d(t) \psi_{a(i,\ell)}(au) \,. \end{aligned}$$

Proof: The control system (24) is written as

$$\ddot{x}+f_1(x)\dot{x}+f_0(x)=\sum_i v_i(t,x)g_i(x)+rac{1}{\epsilon}\sum_i u_i\left(rac{t}{\epsilon},t
ight)g_i(x)$$
 ,

and, according to Proposition IV.3, its averaged system is

$$egin{aligned} \ddot{y}+f_1(y)\dot{y}+f_0(y)&=\sum\limits_i v_i(t,y)g_i(y)+\sum\limits_i \left(rac{1}{2}\overline{U}_{(i)}^2(t)-\overline{U}_{(i,i)}(t)
ight)ig\langle g_i:g_i
angle(y)\ &+\sum\limits_{i< j} \left(\overline{U}_{(i)}(t)\overline{U}_{(j)}(t)-\overline{U}_{(i,j)}(t)-\overline{U}_{(j,i)}(t)
ight)ig\langle g_i:g_j
angle(y)\,, \end{aligned}$$

with initial conditions $(y(0), \dot{y}(0)) = (x(0), \dot{x}(0) + \sum_i \overline{U}_{(i)}(0)g_i(x(0)))$. We compute the iterated integrals of the given oscillatory inputs u_i as

$$egin{aligned} \overline{U}_{(i)}(t) &= rac{1}{T}\int_0^T u_i(au,t)d au = 0 \ , \ \overline{U}_{(i,j)}(t) &+ \overline{U}_{(j,i)}(t) = \overline{U_{(i)}U_{(j)}}(t) = rac{1}{T}\int_0^T \left(\int_0^ au u_i(s,t)ds
ight) \left(\int_0^ au u_j(s,t)ds
ight) d au = -z^d_{ij}(t) \ , \end{aligned}$$

for i < j, so that the averaged system reads

$$\ddot{y}+f_1(y)\dot{y}+f_0(y)=\sum_i v_i(t,y)g_i(y)-\sum_i \overline{U}_{(i,i)}(t)\langle g_i:g_i
angle(y)+\sum_{i< j} z_{ij}^d(t)\langle g_i:g_j
angle(y)$$
 .

Next, we examine the definition of the v_i inputs. Note that

$$\overline{U}_{(j,j)}(t) = rac{1}{2T} \int_0^T \left(\int_0^ au_j(s,t) ds
ight)^2 d au = rac{1}{2} \left(j-1+\sum_{\ell=j+1}^m (z_{j\ell}^d(t))^2
ight)$$

and therefore

$$egin{aligned} &\sum_i v_i(t,y) g_i(y) = \sum_i z_i^d(t) g_i(y) + \sum_{i,j} \overline{U}_{(i,i)}(t) lpha_{ij}(y) g_j(y) \ &= \sum_i z_i^d(t) g_i(y) + \sum_i \overline{U}_{(i,i)}(t) \langle g_i : g_i
angle(y) \end{aligned}$$

where we have exploited the property of the functions α_{ij} . In summary, we have shown

$$\ddot{y} + f_1(y)\dot{y} + f_0(y) = \sum_i z_i^d g_i(y) + \sum_{i < j} z_{ij}^d \langle g_i : g_j
angle(y)$$
 ,

with initial conditions $y(0) = x^d(0)$, $\dot{y}(0) = \dot{x}^d(0)$. Since y and x^d are the solution to the same initial value problem, they are identical. Finally, from Proposition IV.3, we conclude that $x(t) = y(t) + O(\epsilon) = x^d(t) + O(\epsilon)$.

Remark VI.2 (Lagrangian systems on manifolds)

Proposition VI.1 can be extended to a large class of Lagrangian control systems and written in a coordinate-free setting within the so-called affine connection formalism [26], [54]. Let q be the system's configuration on the *n*-dimensional manifold Q, and let $\{\Gamma_{bc}^{a}, a, b, c \in \{1, ..., n\}\}$ be the n^{3} Christoffel functions associated to the system's kinetic energy. Define the operation of symmetric product between the vector fields g_{i}, g_{j} on Qaccording to

$$\langle g_i:g_j
angle^a=rac{\partial g_i^a}{\partial q^b}g_j^b+rac{\partial g_j^a}{\partial q^b}g_i^b+\Gamma^a_{\,bc}\left(g_i^bg_j^c+g_i^cg_j^b
ight)\,,$$

and define the quantity $(\nabla_{\dot{q}}\dot{q})^a=\ddot{q}^a+\Gamma^a_{bc}(q)\dot{q}^b\dot{q}^c$. Then, the Euler-Lagrange equations read

$$abla_{\dot{q}}\dot{q}+f_1(q)\dot{q}+f_0(q)=\sum\limits_i w_i g_i(q)$$
 .

Under the controllability assumption as (A), the result in Proposition VI.1 holds verbatim.

We end this section by considering two examples.

A second-order nonholonomic integrator

There are many interesting dynamical extensions of Brockett's nonholonomic integrators [55] (see the discussion in [43]). We consider

$$\ddot{x}_1 = w_1\,, \quad \ddot{x}_2 = w_2\,, \quad \ddot{x}_3 = w_1 x_2 + w_2 x_1\,,$$

and note that this system fulfills the controllability assumption (A). We design control inputs to track a desired trajectory, $(x_1^d(t), x_2^d(t), x_3^d(t))$, following Proposition VI.1,

$$w_1 = \ddot{x}_1^d + rac{1}{\sqrt{2\epsilon}} \left(\ddot{x}_3^d - \ddot{x}_1^d x_2^d - \ddot{x}_2^d x_1^d
ight) \cos\left(rac{t}{\epsilon}
ight) \ , \quad w_2 = \ddot{x}_2^d - rac{\sqrt{2}}{\epsilon} \cos\left(rac{t}{\epsilon}
ight)$$
 (25)



An illustration of the performance of these controls is shown in Figure 2.

Fig. 2. Tracking for the modified nonholonomic integrator with the controls defined in equation (25) and with $\epsilon = .05$.

A PVTOL model

We consider the model of a simple planar vertical takeoff and landing aircraft model based upon that of [56] with added viscous damping forces; see Figure 3. We parametrize its configuration and velocity space via the state variables $(x, z, \theta, v_x, v_z, \omega)$. We let xand z be the horizontal and vertical displacement of the aircraft, and θ be its roll angle. The angular velocity is ω and the linear velocities in the body-fixed x (respectively z) axis are v_x (respectively v_z). The equations are written as:

$$\begin{aligned} \dot{x} &= \cos\theta v_x - \sin\theta v_z & \dot{v}_x &= (-k_1/m)v_x - g\sin\theta + v_z\omega + (1/m)w_2 \\ \dot{z} &= \sin\theta v_x + \cos\theta v_z & \dot{v}_z &= (-k_2/m)v_z - g(\cos\theta - 1) - v_x\omega + (1/m)w_1 \\ \dot{\theta} &= \omega & \dot{\omega} &= (-k_3/J)\omega + (h/J)w_2 \end{aligned}$$
(26)

Control w_1 corresponds to the body vertical force minus gravity, while w_2 corresponds to coupled forces on the wingtips with a net horizontal component. The other forces depend upon the constants k_i , which parameterize a linear damping force, and g, the gravity constant. The constant h is the distance from the center of mass to the wingtip, while m and J are mass and moment of inertia, respectively.

Equations (26) can be written as a second order system in the variables (x, z, θ) and the model fulfills the controllability assumption (A). We design control inputs to track



Fig. 3. Diagram of the PVTOL model.



Fig. 4. Tracking for the PVTOL model with the controls defined in equation (27) and with $\epsilon = .01$.

a desired trajectory $(x^d(t), z^d(t), \theta^d(t))$ as

$$w_{1} = \frac{J}{h}\ddot{\theta}^{d} + \frac{k_{3}}{h}\dot{\theta}^{d} - \frac{\sqrt{2}}{\epsilon}\cos\left(\frac{t}{\epsilon}\right)$$

$$w_{2} = \frac{h}{J} - f_{1}\sin\theta^{d} + f_{2}\cos\theta^{d} - \frac{J\sqrt{2}}{h\epsilon}\left(f_{1}\cos\theta^{d} + f_{2}\sin\theta^{d}\right)\cos\left(\frac{t}{\epsilon}\right), \quad (27)$$

where we let $c=rac{J}{h}\ddot{ heta}^{d}+rac{k_{3}}{h}\dot{ heta}^{d}$ and

$$egin{aligned} f_1 &= m\ddot{x}^d + \left(k_1\cos^2 heta^d + k_2\sin^2 heta^d
ight)\dot{x}^d + rac{\sin(2 heta^d)}{2}(k_1-k_2)\dot{z}^d + mg\sin heta^d - c\cos heta^d, \ f_2 &= m\ddot{z}^d + rac{\sin(2 heta^d)}{2}(k_1-k_2)\dot{x}^d + \left(k_1\sin^2 heta^d + k_2\cos^2 heta^d
ight)\dot{z}^d + mg(1-\cos heta^d) - c\sin heta^d. \end{aligned}$$

The simulations are run with $m=20,\;J=10,\;h=5,\;k_1=12,\;k_2=11,\;k_3=10,\;g=10$



Fig. 5. Illustration of the tracking errors for the PVTOL model at t = 10 with the controls defined in equation (27).



Fig. 6. Illustration of the tracking in the horizontal displacement of the PVTOL model with the controls defined in equation (27).

9.8. Figure 4 shows an example of the behavior of the controls (27). Figure 5 illustrates the linear decay of the tracking error. Finally, Figure 6 shows how the convergence to the desired trajectory improves as ϵ decreases.

VII. CONCLUSIONS

We have presented a novel and comprehensive coordinate-free averaging analysis for control systems subject to oscillatory inputs. Based on the analysis, we have developed design methodologies for stabilization and trajectory tracking in certain classes of nonlinear systems. Future directions of research include deriving extensions of these results to the case of higher-order averaging, distributed parameter systems, time-delayed systems, and systems with resonances.

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